

AVIATION WEEK

MAY 16, 1949

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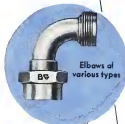
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AVIATION WEEK

Vol. 50, No. 20

May 16, 1949

News Highlights	7	Financial	18
Aviation Calendar	8	Engineering	20
News Digest	11	Production	40
Industry Observer	13	New Products	45
Headline News	15	Sales & Service	53
Letters	17	Air Transport	55
Editorial	66		

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FOR TOP-FLIGHT SERVICE . . .

Big Spring Municipal Airport, Texas



Some of the Big Spring Airport Personnel. Airport Manager Robert J. Cook (left) and two years as Area Supervisor and three years as Captain in the ATC. Flying C-54s plus "over the top."

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AVIATION WEEK, May 26, 1949

University Plan

The "University Plan" of the National Advisory Committee for Aeronautics and the National Military Establishment blueprint a two-year program for aerospace and aerospace-related development. Total cost is estimated at \$1 billion. NACA's mission is to be focused with the first phase of the program over the next fiscal year, which starts in July. This would cost \$77 million and entail 15 transonic wind tunnels for university building and research, three two-foot supersonic tunnels, to be installed at the Ames Laboratory, Langley Field, and the David Taylor Model Basin, and two four-foot supersonic tunnels, at Ames and Langley.

Legislation introduced by Chairman Carl Vinson (D., Ga.) of the House Armed Services Committee and Chairman Milford T. Tamm (D., Md.) of the Senate Armed Services Committee would authorize a \$111 million segment of the program, including \$140 million in student construction, on a USAF engineering development contract, likely to cost upwards of \$750 million. Early construction action is both the House and Senate is expected.

Fitterstein's Charge

United Air Lines President W. A. Fitterstein accused members of the Senate Interstate and Foreign Commerce Committee with his report that no member of the Civil Aeronautics Board, or its staff, charged by law with checking safety financial operations, "has even consulted the thousands of United's doors to study for himself the methods as an airplane to produce the most efficient results."

"No one in the CAB has the first-hand knowledge of what goes on behind the scenes at CAB."

The committee's chairman, Sen. Edward J. Burke (D., Colo.) commented, "That is one of the most startling statements presented" at these hearings on the economic condition of the air transport industry. Fitterstein claimed that "the entire" CAB committee are based entirely on statistical data and "they have helped to obscure the reality of an airline."

Hidden Arms Costs

The modest \$1,113 million, set by the Administration for a first-year, semi-annual Europe program to implement the North Atlantic Pact in only a vague

Johnson's Errors

Aviation observers who said Defense Secretary Louis Johnson's recent speech to the U. S. Chamber of Commerce are wondering whether his guest writer used a lecture on accuracy.

Johnson said these were no Boeing B-47s on order. U. S. Air Force has admitted that only 15 on order. Indications are that contractors have been made for some more although USAF is reluctant to discuss the details because of an operations audit.

Johnson also misled the extent of Boeing's contracts on the B-47s. Shortly before the Johnson speech was delivered Boeing executed three of its major sub-contracts on the B-47 with Martin, Curtiss-Wright and Bell and moved the work back to its Seattle plant.

First production model of the B-47 is now moving down the line at Boeing's Wichita plant.

indication of the size of the program contemplated.

Chief of Staff Gen. Omar Bradley, and Secretary of Defense Louis Johnson, budgeted by Secretary at hearings on the North Atlantic Pact, have conceded that the \$1,113 million, might represent only the cost of re-equipping and transporting equipment now in storage.

Congressional sources anticipate that the actual cost of the equipment will amount to many times \$1,113 million.

Target: Landis

Favorite target of airline officials

remains before the Senate committee investigating industry losses is former CAB Chairman James M. Landis.

United Air Lines President W. A. Fitterstein and he later the approach the recent chairman, Joseph J. O'Connor, Jr., and other CAB members are seeking to attempt to "clean up the mess they inherited."

Fitterstein declared that for 18 months under Landis' direction and control turned to the CAB. This he continued, was the result of "a brilliant but dominant mind endeavoring to hold the reins of his air transport industry without a practical understanding and appreciation of the problems the industry faced."

Landis has announced a three in the sale of many industry executives since he left CAB in January 1948, when President Truman failed to reappoint him. Air Transport News Executive Vice President Robert Kumpke told the Senate committee that Landis is "largely responsible" for permitting non-scheduled lines to violate the law.

Landis previously had told the committee that on such lines and all-gate operators represent a valuable symbol for determining proper costs for regular airlines.

Navy's Needle

U. S. Air Force is showing considerable reluctance to put its General B-36 bomber against Navy fighters as suggested in American War, May 21.

Navy has made a formal request to USAF for use of a B-36 in testing of the defensive capabilities of the Navy's Chance Vought F4U powered by a Pratt & Whitney R-2800 piston engine and the McDonnell Douglas powered by two Westinghouse J34 jet engines. Navy believes both these fighters are capable of successfully attacking the bomber could B-36 at altitude around 40,000 ft.

Although USAF now has two full groups of B-36 bombers in service it has ruled out answering the Navy's request.

Baker Problems

Cancellation of the supercharger to gather with sharp cutbacks in fiscal 1950 development funds have forced Navy Baker to stall design and development of multi-engine aircraft in the 75-100,000 lb gross weight class. But Baker is fighting the situation as a challenge to design aircraft in training the bench, armament, fuel and radar equipment leads into smaller packages.

Publicists of air attack and landing operations will be watching for the changes in the carrier rather than the airplane. Back strengthening of current B-36's mission is one part of the picture. Major problem here is better the right combination of aerodynamic characteristics in the new aircraft design.

Basic ingredient of the new big carrier aircraft is rocket power, both for takeoff and in-flight use. The Douglas A2D and North American A2F variants are currently being re-evaluated to determine if minor modifications such as rocket power and more free air quality can for Midland and Essex-class carriers operates particularly in landing.

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NEWS DIGEST

DOMESTIC

Republic Aviation Corp. XI 91, jet model fighter, made its first flight at Mass. AFB, Calif., with company chief test pilot Carl Belinger as the controls. Flight was made with only the General Electric J-49 jet engine, the rocket engine not having been installed.

Lockheed Aircraft Corp. announced sale of four 749 Constellation to South African Airways. Constellation sales total of the year total 179 to 11 airlines. Dr. Hugh Evelyn, director of structural research of NACA, will report in Washington on May 29 the 57th Wilbur Wright Memorial Lecture to be delivered in London April 28 before the Royal Aeronautical Society. The meeting, sponsored by the Washington section of IAS, will be held in the lecture Dept. Auditorium at 8 pm.

Copier project named for a 100-ft.-long closed circuit was chosen for a Sikorsky S-52-1 which averaged 122.77 mph between Milford and Westbrook, Conn. It is a new craft, and some pilot Harold Thompson, which previously set mark of 129.616 mph over 10-mile course at Cleveland (American Wings, May 9).

First helicopter port in New York City will be officially opened May 18 by Metropolitan Airports Corp., formerly located at Teterboro Air Terminal, N.Y. Company has leased from the city per 41 on the East River. Operating permit stipulates that helicopter can fly only around perimeter of the city.

FINANCIAL

Curtis-Wright Corp. reports estimate dated last week of \$779,504 for this month ended May 31 on sales of \$77,511,777. Loss reflected retroactive price decreases of \$425,000 applicable to shipments of Wright Aeronautical Corp. in 1948. For same period Wright Aeronautical reports loss of \$1,091,000 on sales of \$15,347,042.

Aerobac Industries-Laboratory at ports 1948 operations totaling \$1,815,562, with 38.5 percent of work being for the Government. Employment was up nearly 100 from the preceding year, to a total of 374.

FOREIGN

Island Airways set scheduled service between Ireland and London. Company has one DC-4, fitted to carry 40 passengers in four compartments.

Isle plans to spend nearly \$5 million this year on airport installations.

INDUSTRY OBSERVER

Lockheed has completed the prototype F-93, supersonic jet fighter designed for immense speed. Major changes of F-94 over F-80 series, more pointed nose, fuselage swept back wing and increased power. Initial test flights will be made at Muroc by test pilot Tom LeVier.

U. S. Air Force is to take delivery on first Lockheed F-94 prototype this month. This is night fighter version of 11-40 two-seater jet trainer fitted with special lightweight radar designed by Hughes Aircraft Co. F-94 will be used primarily as night fighter trainer although National Guard squadrons will be equipped with them for tactical use.

Work for North American F-85 to pop back into Air Force procurement program for fiscal 1949-50. Major modifications of F-86 incorporated in F-85 are shift from nose in order to fuselage inlets and design of pointed nose on F-87. XP-93 prototype is scheduled for completion in August. Meanwhile F-86 production has increased to a point where both the First Fighter Group at March Field and the 56th Fighter Group at Selfridge are replacing their F-86s for the supersonic F-86.

Air France will buy an additional Lockheed Constellation for delivery this summer. New transports will be paid for out of ECA funds. ECA claims the Lockheed purchase will be final ECA-financed purchase of U. S. aircraft for foreign airlines until completion of a detailed survey by all Marshall Plan countries in their aircraft requirements and publications for them. Meanwhile ECA will continue to finance purchases of U. S. made aircraft parts and accessories to maintain American transports now operated by foreign airlines.

Republic's successor Air Force a swept wing version of the F-84 Thunderjet. Meanwhile XP-96, powered by General Electric J-47 jet engine and line solid fuel Avon rockets, will function primarily as a research aircraft for USAF exploration of high speed performance data on the ramjet-powered wing.

Pratt & Whitney is now running its PT-2 jetpropeller (early development for Navy) as a ground test stand and plans to install it in use of specially built up B-17 for flight testing. Both Allison and Curtiss Wright used the B-17 nose modification for testing their experimental jetpropellers.

Watch for increasing commercial interest in Pratt & Whitney variable discharge turbine (VDT) propeller. All of the present generation of military transports are powered by PW-Wasp Major engines, but not so the VDT engine. Switch to VDT propeller would offer either a 20 percent climb in fuel required to cover any given distance or a 20 percent increase in range with same fuel. Boeing Stearman and Fiesch SE-201 now powered by Wasp Major may be first commercial transports to switch to VDT.

North American is now buying out production models of the A-1J, three-engine Navy attack bomber at its Downey, Calif., plant. A-1J at first a low speed heavy carrier-based attack bomber. It is powered by two Wasp Major engines and an Allison J-35 turbojet engine. Navy has 48 of this type on order.

British helicopter passenger flight between London and Paris was made recently with a Westland-Hispania S-51. British-built, American-designed, the helicopter carried two passengers and pilot from a London garage to the Place des Invalides in downtown Paris in 2 hr. 20 min.

Financing factor in USAF decision to lease B-36 bomber groups from 15 to 30 planes was the maintenance experience of 7th and 11th Bomb Groups, the first equipped with Convair engine bomber. These groups reported that anticipated maintenance difficulties had led to maintenance that nearly twice the number of planes could be handled without any additional facilities in ground personnel. Some re-equipment of aircraft types is required however since the B-36 requires more of some types such as electrician and less of others.

Sad Plight of Transport Builders

Stratocruiser, Convair-Liner and 2-0-2 projects all tortured to big losses. But DC-6 program shows profit.

Final testimony on the expense of introducing a commercial transport is revealed by the 1948 annual report of the Boeing Airplane Co.

Boeing now estimates that it will lose \$10,000,000 on its commercial Stratocruiser program. Net savings of \$1,758,685 or \$1.58 per plane for 1948 was after a gross of \$7,350,000 at the estimated loss incurred by the company on the Stratocruiser.

Personnel, remaining \$3,300,000 will be recorded against 1949 results. This total excluding loss is equivalent to more than \$6.00 per plane after Federal taxes.

Profit Expected—Boeing fully expected to realize a profit on the Stratocruiser project when it was first launched in 1945. For one thing, much of the basic engineering development had been previously accomplished on the military version. Further, the company anticipated its commercial program with tailfins, which are unusual for a manufacturer.

Unlike other aircraft builders, Boeing executed substantial deposits and progress payments on its transport order. More than seven purchase payments, such as excelsior clauses were incorporated in its contracts.

From orders for 51 planes were received from four European countries and two foreign firms. Deliveries were first expected to start in 1947. Various modifications and compliance tests actually delayed the project. A major slide of the company's Seattle plant last year further perturbed delivery schedules.

The first planes were delivered in January of last year. Trade reports have indicated that because of delivery delays and rising costs, at least one customer attempted to back its purchase contracts with Boeing.

However, Boeing counterattacks were found too uncloud to permit any real release.

1948 Payment Small—As of Dec. 31, 1948, Boeing showed accumulated charges on its Stratocruiser program at \$68,565,692, after the \$7,350,000 credit.

As a partial offset, company had received total advances of \$24,797,019 from its customers. A year earlier, ad-

vances amounted to \$24,780,091, indicating hardly any progress payments during 1948.

As of Dec. 31, 1948, the advance on this business totaled \$18,590,199 which reflects the actual progress and early delivery dates anticipated by the company.

So the 1948 year-end, Stratocruiser backlog suggested \$45,194,000. To finance this program, Boeing was forced to issue bank loans amounting to \$15 million. The management anticipates that the bank loans will be repaid in full this year as the Stratocruiser are delivered.

The cost of its commercial program may have been a major factor in forcing the company to seek additional payments from the government, in addition to progress payments, to help finance its military backlog, last estimated at around \$274 million.

Loss May Be Larger—Until the present Stratocruiser backlog is completed and every plane is in actual service, there is no assurance that the loss on this project will be confined to the current estimate. Experiences of other commercial transport builders must be pondered that other costs, such as modification charges, may be absorbed before the manufacturer's responsibility is ended. By the same token, if Boeing succeeds in adding additional planes, its present projected loss may even be converted into a profit.

A potential additional slide caused a setback as a result of the British contract with Scandinavian Airlines System (SAS) previously had ordered to Stratocruiser. The Scandinavian contract, which held a contract in four planes (two-thirds of the order to the British) is a reported profit through the generous workings of foreign exchange. The British are reported to have paid for these four planes in blocked sterling and thus avoided the outflow of American dollars. Interestingly enough, the funds for the original British order of six Stratocruisers are being provided by the American taxpayer through the Marshall plan.

The final accounting on the Boeing Stratocruiser program may not be sent in until all of the results are in a few years from now.

DC-6 Example—Like Douglas DC-6 program started out with a huge loss but that was finally converted into a profit.

For example, in its 1946 annual report, Douglas wrote off \$5 million in experimental and development costs, in addition to the \$5 million lost on the delivery of six DC-6s made during that year. Under the company's system of accounting, the delivery of the DC-6s earned a much higher percentage of the development and the engineering costs.

Losses on DC-6 program continued throughout 1947 with the company providing a total of \$4,001,628 at Nov. 30, 1947, in the estimated additional cost for modifications and other purposes as its delivered DC-6s. This was previously successful for Douglas reporting a loss of \$2,100,379 after tax carryback credit for 1947.

The company's financial results changed materially for the better, however, when it reported net income of \$7,329,256 during the fiscal year of 1948.

According to the Douglas management, the conditions arose "from the effect of the DC-6 program. Under this accounting treatment, there were heavy charges against income in 1946 and 1947 for the development of the DC-6. DC-6 production costs were reduced considerably from those incurred against 1947 deliveries, and 1948 sales were made at a modest profit compared with a large loss in 1947."

Less Fervent Example—Less fervent in its commercial transport ventures are both the Consolidated Vultee Aircraft Corp. and the Glenn L. Martin Co.

Convair, while it had first orders for 175 of its 244s, has had a disappointing series of losses on this project. At July 31, 1947, the company provided for \$16,100,000 as a loss on this program at that time. At Nov. 30 of the same year, the management estimated an additional loss of \$11 million would be incurred by the company if more planes were not sold.

Moreover, at Nov. 30, 1948, company provided a reserve for warranty amounting to \$1,087,000 to cover charges against warranties issued against deliveries at this place.

The Glenn L. Martin Co. took a loss of \$22,089,497 on its 2-0-2 program during 1947. A further loss of \$17,718,080 was taken during 1948, though the same method—a write-down of its work in process inventories covering this aircraft.

On the whole, recent experiences of the aircraft builders in developing transports for the commercial market has not been very happy.

—Selig Abraham

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GIN TRAIL: GROUND, position of aircraft within blocks is indicated by dashed white lines.



IN THE ATR pilot gets view from this.

Visual Signals Used in Traffic Control

New fixed airspace separation system, which operates without voice communication, gets intensive tests.

By O. S. Field*

An experimental air traffic control fixed airspace separation system, which gives visual indications to a pilot and requires no voice communication, has been developed by General Railway Signal Co. It has been in test operation for more than two years with promising results.

Developed in cooperation with Westinghouse and the Air Materiel Command, the system is one of the first attempts to replace voice communication in air traffic control, and is designed to meet the requirements for a "full scale" block system established by Special Committee 13 of the Radio Technical Commission for Aeronautics.

In about 150 flights with two C-47s over an experimental runway near Rochester, N. Y., the system has been extensively tested by flight personnel of the Air Transport Command, and the Civil Aeronautics Administration. The system detected double occupancy of the blocks as matter when the aircraft were located within the double-occupied volume element. This was true under adverse conditions in which the two aircraft were within 100 ft. of each other.

In addition, relative positions were fixed adequate to separate the system even when the aircraft was at 12,000 ft. directly over the ground station. The

positions of the segment boundaries were visible to within a few hundred feet during the entire testing period.

Automatic and fully automatic air traffic control. In the first place, the system must continuously, and without human intervention, sense elements that will maintain safe separation. In the second place, when subjected to human intervention (such as a pilot's control) the system must be capable of moving volume elements.

In the discussion of the ultimate system of air traffic control, frequent reference has been made to two methods. These are positively referred to as fixed and moving block systems. In a moving block system, means are provided for establishing a zone of influence around each aircraft. These means move with the aircraft and serve as barriers for protecting columns.

On the other hand, the blocks of airspace assigned as a fixed airspace separation system (fixed block system) are seen fixed with respect to the terrain and the aircraft fly through them in succession.

Fixed Block—In a fixed airspace separation system, air separation is maintained by dividing the controlled airspace into a pattern of volume elements, or cells, and permitting the occupancy of a given volume element by only one aircraft at a time.

In the case of an airline, the pattern of volume elements is set up by first dividing the line into segments by means of vertical planes transverse

to the runway. The lengths of the segments may be ten or more miles, depending upon traffic density. The segments are then subdivided into altitude layers by a family of horizontal planes spaced normally 1000 ft apart. Finally, vertical planes are spaced which cut across the four boundaries.

A volume element is thus a block of space bounded on each end by two receiver segment boundaries, on its top and bottom by the planes that define the altitude boundaries, and on the top and bottom by horizontal planes spaced 1000 ft apart. The dimension of a typical volume element in a high traffic density region would be ten miles long, five miles wide, and 1000 ft. thick.

The geometry may be visualized by imagining the controlled airspace to be built of volume elements in the same way that a wall is built of bricks in the special case in which each brick is laid directly above the one beneath it.

General Control—To connect traffic control, an integrated control system is established on the ground. This control system must perform three vital functions. It must be able to:

- Detect continuously which volume elements are occupied.
- Analyze occupancy conditions and serve as a set of safe, non-conflicting clearance for all aircraft under its control.
- Transmit these clearances to the specific aircraft to which they apply.

In addition, the system must be arranged so that in the performance of the second function it recognizes both pilot requests and in the assignments of expeditious traffic movements.

The detection of the occupancy of a volume element is a vital function and

must therefore be accomplished by the most direct and straightforward means possible. Likewise, in the case of full automatic operation, the relay of occupancy information must be kept to a minimum.

The heart of any traffic control system is the mechanism that supplies traffic conditions and serves as a set of clearance elements. In the design of such equipment, several principles are essential:

- Clearance must be issued at zero latency. This subordinates mechanical elements for human judgment, even the controls of safety responsibilities, and permits him to concentrate upon flow control problems.
- Clearance must be made in a safe and safety must never depend upon the position of human signal positions. The hazards of predicting in view of variations in wind and light conditions are obvious.
- Space for an intended movement must be reserved before clearance is issued. In general, thus, clearance to enter the next volume element along an avenue is a guarantee that all other aircraft have been cleared from that block of airspace and cannot receive clearance to enter it. Highly developed techniques in the form of interrelated relay circuits are already available for accomplishing this function in accordance with full-scale and closed-circuit principles.

The transmission of clearances from the ground to the air must be accomplished in such a way that there can be no danger of an aircraft's accepting any but the clearance intended for it. Automatic repeat-back of the clearance for checking purposes, and the withholding of the clearance display until the check is completed, are necessary.

Furthermore, in the safety clearance for a given aircraft depends only upon the position of the aircraft and not upon its flight number or other identification, it is desirable to add the safety message directly to the volume element occupied by the aircraft.

Rochester System—The present experimental fixed airspace separation system has developed four ground stations located at Rochester, Spencerport, Albion and Livestock, N. Y. These stations are approximately 15 miles apart along the northeast leg of the Rochester relay station and serve to divide the single line service into four segments.

The boundaries of these segments are simply half-way between stations, so that there is one segment approximately 15 miles long associated with each station. A ground wave of the center one of the four stations is shown above. The station on left have both 1000 mc. radio frequency equipment and microwave relays for deriving occupancy and establishing priority for communication with the aircraft. The stations on the right have only microwave relays for the traffic conditions at the Rochester segment.

As shown are three that are connected with transmitting the occupancy conditions in the Rochester segment to the ground station at Spencerport. The station at Spencerport is shown in the background.

As a ground station in ground. A red light means that it is unsafe for the aircraft to proceed at the same altitude across the boundary into the next segment. The pilot must then either hold or descend below the boundary. One element normally occupies, or request clearance to change altitude layers.

Master Control—Although the system is fully automatic and uses safe separation clearances without any human intervention, a central control (photo, page 26) has been provided at the master station to permit the initiation of dispatching or flow control clearances to expedite traffic.

At the experimental system occupies four segments and four altitude layers within each segment, there are sixteen volume elements shown on the master control panel in the Rochester station. The ground system, of course, is set located in the four altitude layers and four longitudinal segments of the experimental installation but can be expanded as needed to meet actual or new operating requirements.

The board shown is a schematic representation of the overall traffic picture. Each volume element is represented by a line on which an indicator light and a label are mounted. These lines are arranged in stacks of four, which represent the segments, and each segment is labeled with the name of the station with which it is associated. When a volume element is occupied, the corresponding light on the board is illuminated. Thus a condition has a convenient picture of the traffic before him at all times.

To avoid flow-control clearances to a given aircraft, the label associated with the volume element occupied by that aircraft is used. For instance, by giving the label a quarter turn clockwise, it was a clearance to proceed 1000 feet in the next volume element. It was an unsafe clearance to proceed in the same direction that clearance issued previously, or otherwise, are transmitted only after the safety of the intended movement has been checked automatically by the interrelated relay circuits and then only after the required spacing has been assured. An unsafe clearance is not transmitted.

Display Indicator—The pilot of an aircraft in the system receives clearance on the cockpit signal indicator (photo, page 26). The turning signals are selected by means of the four lights assigned on the basis of a code about the light label "Clearance" or "Occupancy". The lights on the vertical axis of the cockpit are green and those on the horizontal axis are red. It is possible therefore to tell back the color and by position what signal light is being displayed.

A green light is clearance in ground. A red light means that it is unsafe for the aircraft to proceed at the same altitude across the boundary into the next segment. The pilot must then either hold or descend below the boundary. One element normally occupies, or request clearance to change altitude layers.

With proper flow control the occu-



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serve of a red radiation is rare, and when it does occur holding will not seriously result, since a change in altitude or lane will usually permit the setting up of a safe path around the interfering traffic.

The position of the aircraft relative to the segment boundaries is displayed on the meter at the bottom of the window. When the correct boundary is ten miles or more away, the needle is fully deflected to the right. As the boundary is approached, the deflection of the needle decreases toward the first division at the left end of the scale, and as the boundary is passed, the needle gives a sharp back-lash to the right. Failure of the indicator circuit is shown by zero deflection.

► Operation—The meter is required for clearance to proceed, to descend, or to enter the runway visually. The pilot uses the request knob located to the right and above the boundary meter. Clearance to proceed, for example, is requested by turning the knob to the "Accept" position and pressing it.

If the clearance is safe, clearance is granted within two seconds and the "Accept" light with the arrow pointing upward goes on. Descent and entry are handled in a similar way. A light is provided below the request knob to permit the cancellation of requests. The same lights are used to display flow control clearance signifying on the ground.

In case violation of a clearance results in the occupancy of a volume element by more than one aircraft, all the aircraft concerned are notified at the emergency by the lighting of these double occupancy lights. Action is then taken in accordance with prescribed procedures.

► Field Installation—For the sake of flexibility during the initial testing period, the experimental field stations were mounted in 14-ton trucks. An interior for communicating with the aircraft was fitted located on the platform at the rear of the truck. Engineers are the mounting and securing elements of the VHF communication links between stations, and two parabolic reflectors near the top of a 100-foot mast belong to the ultra-high frequency synchronous sonar field.

For simplicity in describing the operation of the system, it will be assumed that the stations are equally spaced and that the segment boundaries occur midway between stations. A further timing element at the Rochester station provides timing pulses at the rate of 1500 per second that are passed from station to station over a sharply band-passed VHF synchronous link.

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AVIATION WEEK, May 16, 1949

Evaluate Nitromethane for Rockets

Although suitability of this liquid monopropellant is controversial, its properties show much promise.

Many studies have been conducted before the rocket engine can be operated as simply and reliably as the compressing engine, but it is worthy of these studies are closely associated with the propellant themselves. Consequently, the search for better propellants is being pushed throughout the field of rocket research.

Nitromethane, considered the most promising rocket propellant of the jet-propellant group, has been under study at the Aerojet Engineering Corp., Azusa, Calif.

Suitability of this fluid is being vigorously argued among scientists and engineers in the rocket field. The question is whether the explosives of nitromethane will have a serious effect upon its general acceptance as a propellant will not be finally answered by any experiment, but more statistical evidence and much more developmental experience with actual engines will facilitate a decision.

► **Propellant's Background**—Characteristics and advantages of this fluid, detailed below, were disclosed last month at the American and Air Transport Meeting of the Society of Automotive Engineers in New York, by Dr. Fritz Zwicke, director of research at Aerojet and C. C. Kien, chief engineer, liquid engine department.

Theoretical and experimental research study of nitromethane was begun at Aerojet in 1943. Basic properties of nitromethane as a rocket propellant were investigated, with consideration given to use as a monopropellant and its combination with auxiliary propellants and oxidizers.

These investigations were performed in tests of rocket engines of various sizes and in experiments with gas generation of the type used in comparison with gas turbines.

In 1945 an engineering project was initiated to study the potential use of nitromethane as various propellant

monopropellants. This work, has definitely proven the feasibility of using this fluid as a monopropellant.

Total amount of development work done to date is small when compared to that done on the conventional liquid propellants, hence, development of nitromethane rocket engines and engines is still in an extremely early stage.

► **Propellant's Distinctions**—In general, liquid rocket propellants may be classified as to propellants as monopropellants. Nitromethane combination usually consists of two liquids—an oxidizer and a fuel. These are introduced separately into the combustion chamber where, upon contact, they react explosively.

Monopropellants consist of a single liquid propellant, under controlled conditions, the ability to decompose in the rocket combustion chamber, liberating heat and gaseous products.

With bipropellants, the oxidizer and fuel are introduced separately into the combustion chamber and react upon the design of the jet-propellant system.

At present there are three principal oxidizers which have been tested with nitromethane: liquid oxygen, nitric acid, and hydrogen peroxide. Each has been investigated in connection with a study of both. Some of the principal liquid propellant combinations are shown in Table 1 below.

► **Bipropellant Performance**—Specific impulses to be expected from these bipropellants are given in Table 2. It is apparent that, with the exception of the liquid oxygen-liquid hydrogen combination, theoretical specific impulse obtainable from liquid bipropellants ranges from approximately 200 to 275 seconds. The corresponding range of effective exhaust velocities being 6850 to 9220 ft./sec., respectively.

A disadvantage which must not be ignored in the design of the propellant is an important factor which detracts,

Table 1—Liquid Bipropellant Systems

	Liquid Oxygen (LO)	Nitric Acid (HNO ₃)	Hydrogen Peroxide (H ₂ O ₂)
	OXIDIZERS		
		Alcohol (C ₂ H ₅ OH)	Hydrocarbon (N ₂ H ₄)
FUELS	Gasoline (C ₈ H ₁₈)	Hydrocarbon (C ₁₀ H ₁₈)	
	Ethyl Alcohol (C ₂ H ₅ OH)		
	Liquid ammonia (NH ₃)		

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to a certain extent, size and weight of the jet propulsive system. A desirable performance criterion for certain applications is the product of specific impulse and average density (actually, typically, grossly) of the propellant, designated by the symbol L .

If the density-specific impulse is used as the performance criterion, choice of bipropellants is limited to a range of 175 to 167 L. (if the combination of liquid hydrogen plus liquid oxygen is excluded).

This narrow range of values is not great enough to serve as a basis for the choice of any particular combination of the physical and chemical properties of the oxidizer (and of the fuel in certain cases) as when one selects.

► **Oxygen Factor**—Evaporation and decomposition have shown that, from the viewpoint of peroxidizability, the physical and chemical properties of the oxidizers are more decisive in judging suitability than are small variations in the density impulse or the specific impulse.

Liquid oxygen is the oxidizer used and solid is not popular. A variety of uses of rocket motors employing this oxidizer have been reported successfully. But fewer liquid oxygen powerplants than that used in the German V-2 rockets.

Principal disadvantages are its low density and storage problems. It boils at -183° F., must be stored in specially insulated containers to prevent very rapid evaporation. Despite such precautions, there is an inevitable loss due to evaporation during storage.

The high vapor pressure at ambient temperatures (which are almost invariably above the low boiling point) complicates its handling in pumping units, and the low temperatures which exist period during storage lessen the choice of container materials. Liquid oxygen is dangerous to handle, and in contact with the skin produces effects similar to frostbite.

Furthermore, although it and fuel

used with it are relatively safe when stored in moderate conditions, they can ignite a violent fire and explosion based whenever there is a possibility of contact, a condition which will occur, for instance, in the event of a crash landing.

► **Nitric Acid**—Second important oxidizer in nitric acid. This is used in several modifications such as acid fuming nitric and containing up to 65 percent oxygen nitrogen dioxide, or a white fuming acid and containing no more nitrogen dioxide.

Considerable experience with the various modifications of the oxidizer has been obtained in this country.

Nitric acid is less hazardous with respect to the temperatures which must prevail during loading and storage than is either liquid oxygen or hydrogen peroxide. However, because of its corrosiveness, containers employed for nitric acid must be made of stainless steel.

If proper handling equipment is used, its transfer from containers to the tanks of the jet propulsive unit involves no real difficulties.

In addition to corrosiveness, principal objection to nitric acid as an oxidizer is the toxic fumes it gives off and the fact that in immediate service bias occurs upon contact with the skin.

As in the case of other bipropellant combinations, nitric acid propellants represent a fire hazard in the event of a crash, especially since it is normally used with a fuel spontaneously combustible on contact with the oxidizer. Thus, nitric acid propellant combinations are perhaps better suited for remote powered and smaller devices than for piloted aircraft.

► **Hydrogen Peroxide**—In high concentrations, this substance has also been employed as an oxidizer in bipropellant rocket motors which have been operated successfully. (The propellant of the German ME-163 interceptor is a notable example of this application.)

Hydrogen peroxide does not have the same storage requirements as either liquid oxygen, but it is more dangerous to handle. And its use in the rocket field is so new that the operating data essential to its widespread application are still incomplete.

Since the field will decompose under certain conditions, it may be classified as an explosive. Thus small amounts of impurities such as those present in dirty containers or pipes, and also inadequate heating, can start self-decomposition of the hydrogen peroxide and explode the container. Explosions of this sort have occurred in a number of merely exposing the container to sunlight. Hydrogen peroxide solutions of high concentration (above 57 percent) are sensitive to shock, and may be exploded by perforation.

In addition, the third chief concern is a fire hazard, since it will initiate the combustion of oxidizable materials (such as wood) upon contact with them. This has been caused during transport by leakage from containers or supposedly empty drums which have been allowed to rest on a wooden floor.

Consequently, in some degree, in each of these oxidizers. This characteristic creates a difficult problem in the selection of construction materials, as noted above, and also limits the service life of parts, requires their frequent replacement and necessitates extensive servicing operations.

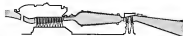
► **Nitropropellant**—Detail—in contrast to the bipropellants, a monopropellant consists of a single liquid which has the properties, under certain restricted conditions, to decompose with the evolution of heat and gases. To compete with bipropellants on the basis of performance, total heat release (the work of the monopropellant must be of the same order as that of the bipropellants).

A paradox is immediately presented in that the superior value of the best oxidizer must also be of the same order as that of a high explosive. Hence, a

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Table II—Calculated Performance Characteristics of Liquid Bipropellants

Oxidizer	Fuel	Mixture Ratio		Exhaust Velocity (ft./sec.)	Specific Impulse (sec.)	Density (lb./cu. in.)	Chamber Temp. (°F.)	Mean Prop. Th. (in.)	Mass Flow Rate (lb./sec.)	Specific Heat Ratio
		Proportion	Ratio (Oxide/Fuel)							
Liquid Oxygen	Gardol (Ref. 1)	180	2.1	7700	142	235	5470	21.7	1.21	—
Liquid Oxygen	180% Ethyl Alcohol (Ref. 1)	300	1.2	7610	141	221	5160	—	—	—
Liquid Oxygen	Ammonia (Ref. 1)	300	1.4	5230	115	249	4950	—	—	—
Liquid Oxygen	Methane (Ref. 1)	300	1.0	7020	136	246	5100	—	—	—
Liquid Oxygen	Liquid Ethanol	300	1.0	7140	138	260	5120	—	—	—
Liquid Oxygen	100% Ethyl Alcohol	300	1.1	7500	139	233	5060	22	1.21	—
Nitric Acid	100% Water (Ref. 1)	180	3.0	7990	151	367	5020	25	1.22	—
Nitric Acid	Alcohol (Ref. 1)	—	—	—	—	—	—	—	—	—
White Phosphorus	Terbutyl Alcohol (Ref. 1)	180	1.9	6810	124	293	3820	—	—	—
Hydrogen Peroxide	Hydrazine Hydrazine (Ref. 2)	150	—	—	100	—	2000	16.0	1.2	—



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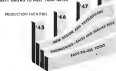


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decomposition temperature is in the order of 500 F. Because of the spread between these two temperatures, the fast is safe to handle under ordinary conditions. Temperatures of the range of 500 F. are not inherently an accident and can be avoided with care.

Also, the liquid must be well confined to prevent its boiling off before the fastener is installed.

Thus, it can be seen that present temperature problems in connection with non-aqueous fasteners, or at least can be handled by proper design and use with relatively large margins of safety.

Shock Sensitivity—This is another property of such materials and must be considered as important to temperature sensitivity in the practical use.

Shock may affect adhesion by its tearing complete or partial decomposition, depending on the conditions of environment and the severity of shock. However, the evaluation of shock is most difficult and is usually made on a statistical basis. Some general observations of this nature are:

1. Non-aqueous will not deteriorate in thin-walled containers where vibration is attempted with a No. 3 blasting cap.

2. It will not deteriorate in heavy-walled metal containers upon striking the ground with an extended distance from 5000' after being dropped from an airplane.

3. It will deteriorate in heavy-walled containers (steel) when subjected with limited charges of tetryl or other high order explosives.

From these examples, it can be seen that under normal circumstances, the material is extremely stable from a stand point of shock. However, in all tests to determine shock sensitivity in which detonations have been initiated, there is a question as to whether the initiation was accomplished by shock alone or by a combination of shock and the liquid heating caused by the reaction.

Explosion Types—Inherent decomposition of non-aqueous in bulk form may be by extreme temperature or shock conditions may result in one of two types of explosion.

Most serious of these two explosions would be a complete detonation of high order and complete propagation of the detonation through all lines and fittings in the system. The other type is a local, low energy decomposition of a small quantity within the bulk.

The high order type of detonation had been produced only in experimental testing where the test sample was subjected in particularly severe conditions. There is much statistical evidence available to show that none of the types of mishaps which can occur in rocket systems can impose the extreme conditions necessary to initiate a complete detonation.

(Continued on p. 17)

Table III—Properties of Commercial Nitromethane (Ref. 4 and 5)

Chemical Composition	CH ₃ NO ₂
Molecular Weight (g/mol)	61
Percent of Decomposition	CO ₂ + CO + H ₂ O + N ₂ + H ₂
Heat of Decomposition	1710 Btu/lb
Specific Weight	0.867 lb./cu. in. at 70 F
Vapor Pressure	0.17 psi at 70 F
Viscosity	2.74 cps at 150 F
	1.15 cP at 100 F, 0.70 cP at 70 F
	0.901 lb./cu. in. at 100 F

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Westinghouse hoisting was put to test when asked to design a cargo hoist and this would be dependable under all conditions . . . be completely explosion-proof. The answer is illustrated here. This unit can lift 4,500 lbs. of cargo at 34 feet per minute . . . weighs only 2,000 lbs. The main drive consists of a rectangular cast, stainless steel planetary gear, a speed limiter and a magnetic brake. Because of its long life, no space need be carved by the plane, saving weight.

Lights penetrate heaviest fog

A major problem of the Airlift has been its fog-avoiding fog. To combat this condition, Westinghouse flashing beacon lights are being installed in seven Airlift fields for observation purposes. These lights make visual landings possible in the worst weather conditions. Flashing 40 times a minute, the lights can penetrate the heaviest fog for a distance of at least 1,000 feet. However, the probe flash does not blind the pilot because its apparent duration is so short. The clear or busy days, or clear or light foggy nights, the landing can be reduced.

Safe transformer operation

The "CSP" (Compactly Self-Protecting) transformer—long accepted as the best transformer under all conditions—provides itself against lightning, short circuits and overloads. At the Airlift, 40 "CSP" transformers are being used for approach and other aerial lighting.



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PRODUCTION



ALUMINUM SHEET is held in jaws of stretch press, pulled inward around final and...

Stretch-Wrap Forming at NAA

Manufacturing trend is toward heavier materials for high-speed aircraft.

By Alexander McFarley

INGLEWOOD, CALIF.—A technological revolution in airplane manufacturing techniques is well in its early stage in the systems of J. L. Atwood, president of North American Aviation, Inc. "Today's high-speed planes could not

be produced by methods used by our industry as recently as 10 years ago," Atwood states. "The trend is toward still heavier materials, heavier machine tools and more exacting tool designs in the future."

Greater manufacturing pressures are being faced by the use of the latest



WING SKINS on North American's F-46 are tapered in place with flat cut away material in areas where it is not needed for structural strength.



WARRANTED on the die form and using

materials used to take higher stresses of higher speeds.

"No Second Chance"—You can't make 775 alloys to fit or you can't make many mistakes," Atwood points out.

In making its two principal products (production military planes, the F-46 jet fighter and the F-47 jet bomber), North American has put to more extensive use a "stretch and wrap" manufacturing technique which it first began to use during World War II.

The Inglewood plant has three units of stretch-wrap forming machines, products of the Haffner Company, Redondo Beach, Calif. They range from a machine with 100 tons stretching force down to one with 10 tons of stretch.

The smaller machines are limited to the forming of stronger, other more sheet metal parts and extrusions. The large machine handles sheet metal parts, including skin sections, flaps, augs and wingtips.

Wrap-around-forming process involves first stretching the sheet face of the die to meet the yield point and then wrapping it around the die with a final additional stretch to set the material to the die contour.

Production analysis has indicated that the stretch-wrap process is superior to deep drawing forming because the rate of drawing can be controlled, and because the spring-back of the metal is eliminated when it is pulled beyond its yield point.

The machine holds the material in place at either end, which pull apart for the stretching action, and then pull to uniform the wrapping action.

J. H. (Dutch) Kinschberger, chief man of the North American board has

suggested a novel jaw arrangement for the Haffner machine which is expected to result in a savings of material.

► **Tapering**—Some of North American's other primary manufacturing processes now in use are expected to be supplanted by some specialized equipment in this category in the method of tapering wing skins for the F-46 fighter.

Besides a tapering press, wingstock to simplify where strength requirement is least, the design called for variation in thickness of skin to meet where the stresses were to be transmitted to bulk or ribs.

For example, an area about one-half inch wide along each edge of the skin at the tip was designed to be flatter than the rest of the skin, in order to transmit load to the spar. To get the variation in skin thickness it was elected to weld the excess material from the skin sheet.

Concrete Milling Machine plates were used and the heavy skin sheet a hundred for the refining operation by an interesting vacuum jet lifting device. A triangle shaped frame with three vacuum ports a small vacuum pump and a motor are attached to an overhead conveyor. Cops are lowered against the sheet, at points which remain flat after the milling operation, and the sheet is lifted to the next operation.

Amount of milling required, and amount of material lost in the method of skin production suggests that an alternate process of skin tapering will eventually supersede it.

PRODUCTION BRIEFING

► **Acro Airlines**, machine product subcontractor at Tullahoma, N. Y., plans to open a branch plant at Hickory Field, Fort Worth, to serve Chance Vought at Dallas. Company produces precision tools, dies, jigs, fixtures and experimental and production parts. Donald de Ladt, president, and he would take a drive over to Fort Worth to open the new plant.

► **Faxon Manufacturing Co.**, Cleveland, has scheduled a \$2.5 million expansion program for its new plant at Balfour Creek, Mich. Plant makes compressor blades for jet engines.

► **Scott Aviation Corp.**, Lancaster, Pa., has begun production of an oxygen indicator for use in turboprops, doesn't offer and industrial fuel-oil mixer. The new device is based on an Air Force automatic-control oxygen mask development.

► **South Aircraft Corp.**, Wichita, received a \$3 million contract from the Great American Farm Implement Corp., Chicago, to make several thousand

F.O.B. Less Tax

Faking delivery on Shinkansen cars is going to be easier from here on out.

Before May 1, Boeing Airplane Co. had to deliver its commercial planes in Portland, Ore., across the Washington state line, to avoid a 5 percent sales tax. It made a near 35-minute flight for the price and was worth the tax, saving. Otherwise it was pretty much of a nuisance.

The last state legislature abolished the tax as far as airplanes are concerned, effective May 1. First delivery from Seattle was scheduled for last week.

Each of a new-type core laminator Bochs now is testing for production that is expected to produce only next December. About 2000 additional employees will be hired.

► **National Ribbon Co.**, Buffalo, received

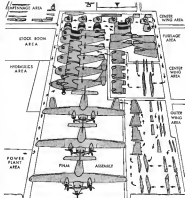
a \$142,450 order for 5000 aircraft batteries from the U. S. Air Force. Batteries are for use in Douglas C-54 transports as well as on the Berlin airlift.

► **United Helicopters Co.**, Palo Alto, Calif., now is testing test two Sikorski 360s per week for commercial sales, and expects to step up production to four per week. About 1000 have been delivered with an additional 60 on order.

► **Canadian Car & Foundry Ltd.**, Montreal, will make spare parts for the Canadian-built F-46 jet fighter at its PT. William plant. Chicago now is making parts in the wartime fighter plant. The F-46 will be built by Canadian Ltd. under license from North American Aviation, Inc.

► **Griffith Bros.**, Los Angeles, is negotiating with the Canadian government for production of its CPN-4, an inconvertible two-seat GCA equipment. The Canadian government will select a Canadian firm to manufacture the equipment under license.

► **Republic Aviation Corp.**, will close its plant Aug. 1-15 to give all employees

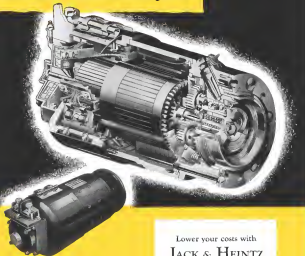


NORTHROP'S RAIDER LINE

Schematic diagram of how Northrop plans to set up production line at its Hawthorne, Calif., plant for the Raider (C-124A) as well transport and Arctic rescue plane as

shown here. Northrop has a GAUF area for 23 Raider. Most assembly line across down the center of the plant with sub-assembly loading at front both ends.

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attach-detach mount permits removal of starter in a few seconds. Many features of simplified construction assure dependable operation and long life. Write for full details today.

two-week vacation with pay. Skeletal maintenance and plant protection crews will be on duty during the vacation shut-down.

WHO'S WHERE

Airproducts division of General Motors Corp., Dayton, Ohio, named Mr. M. Moore general manager. With GM for 26 years, he has been acting general manager since the death of W. J. Blanchard in an airplane accident last December.

Thompson Products, Inc., Cleveland, Ohio, appointed J. D. Wright general manager. He is also vice president and secretary, and has been with company for 16 years. Thompson also promoted four division managers to vice president: Harry D. Bello, Tapes plant, Euclid, Ohio; Leo W. Barnes, special products division, Cleveland; Matt F. Graham, Detroit; and Paul D. Hoffman, who has been in Wall Coast plant, Los Angeles, since 1957.

Standard-Thomson Corp., Dayton, Ohio, appointed C. W. MacNeil director of a new chemical and metalurgical laboratory, now being established by company. MacNeil formerly was associated with Igo Iyer Co., Duquesne, and Wisconsin Steel Works, Chicago. He was an Air Force instructor during the war.

Lackhead Aircraft Corp., Berkeley, Calif., has reorganized its sales staff to bring expert sales credit direct report under of Leonard K. Schwartz, general sales manager in Berkeley. P. K. Yonker, former district sales manager in Berkeley, becomes assistant general sales manager with headquarters in New York City. He also will assume duties of eastern district sales manager. R. H. Adams, former export sales manager, has been named sales representative for Australia, India and the Far East. Although he has long been with the company, Adams will spend most of his time in the field. James B. Boyce with Lackhead has many years in engineering and service branches, has been appointed sales representative for Europe, Africa and the Near East, with headquarters at Hotel Desbrennes, Geneva, Switzerland. John Wagner continues as regional representative for South America, with local quarters in New York City.

G. M. Cannon & Co., Inc., Pasadena, Calif., manufacturers of guided missile and aerospace aircraft instruments, elected Test Admiral Ken de France, USNR, to the board of directors. France is a veteran of the Robert F. Collier Trophy for his extensive contribution to safe and rapid training of aircraft pilots and crews.

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Other manufacturers depend upon outside sources of supply for their valve parts. Thus, slide valves are designed to meet your specific requirements. And they must also be designed to fit into other manufacturers' standard actuator units. This frequently results in a compromise valve—a valve that may be acceptable but is not the best possible valve due to limitations imposed by the actuator unit. Now, compare this with Whittaker motor valves. Here are complete valve assemblies that are designed, engineered, built, assembled and tested as a unit. Link valve bodies and actuator units are individually engineered to meet your specific requirements. They are developed as a unit, made in the same plant, and assembled and tested together. No compromise will ever be made. You have one source of supply, one source of quality and one source of responsibility. Your engineering, purchasing and assembly men know they can be ordered to a minimum. Make this comparison and you'll see for yourself why Whittaker valves are fast choice among the leading aircraft manufacturers the world over.

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Fast with valves that are fast choice in industry

Reversible Driver

Powermatic Close-A-T screwdriver is designed to speed production jobs that require driving many small screws and other fasteners. Small size, 5½ oz. weight, and freedom from torque, combine to make it easy to handle. Made by Cleco division, Rockwell Int'l Co., P. O. Box 2119, Houston 1, Tex., screwdriver is claimed to be only 4 oz. heavy in other tools of similar capacity. It is considered especially useful in driving operators fatigue on jobs requiring continuous operation. Rotated impact mechanism shortens driving time, permitting fingertip control and there is said to be no possibility of tool twisting in operator's hand as "popping out of action" due. Since throttle valve is opened by pressure on bar, variable "torque limiting" is eliminated. Rotation is easily reversed and reversing mechanism of tool unnecessary.

For Radiation Studies

New high-sensitivity thermocouple, Hans-Oberle type, is offered by Fine and Optical Co., Inc., 4405 Bronx Boulevard, New York 46, N. Y. In transparent is claimed to be particularly suitable for thermal radiation measurements involving chopped or modulated radiation at frequencies up to 10 cps. It has active layer surface of 4 mm square, spectral range, with 10% sensitivity, of 0.1 to 15 microns, resistance of between 6 and 10 ohms and dielectric sensitivity greater than 10, obtained for time constant of approximately .05 mseconds. Dimensions of thermocouple case have been chosen to subtend approximately 10 percent of area of 60 mm diameter spherical source when mounted thermocouple placed at its focus. Device is available constructed as uncompensated element or as compensated, and covered with dust-coat, tape covering to absorb gas which will leak into it but which does not rise with of thermocouple case.

Parts Cleaner

Improved Formula 404 for removing paint, sludge and light carbon from aircraft engines and parts and for similar metal cleaning is announced by Kellogg Products, Inc., Box 7817, Terminal Avenue, Los Angeles 54 Calif. Material is claimed to be safe on all metals including cadmium and zinc, and will clean with much contamination. It has low evaporation rate, is used at normal room temperature, and will last from 10 to 140 F. Material is packed in 15 and 15 gal drums and 5 gal cans.

Precise Control—for Intricate Operations



Aeroprops—with Electronic Turbo Propeller Controls

Application of propellers to turbine engines has reproduced many exact control requirements, among them being more precise governing, increased stabilization, and a greater range of operating blade angles.

Now these and other requirements are met by the Aeroprop Control System, which not only fulfills the demands of a turbine engine installation, but also provides automatic synchronization for multi-engine installations. This tested control provides precise automatic R.P.M. reduction, acceleration control which is

unmanned by means of time lags, efficient engine-to-engine, and automatic synchronization preventing withdrawal of one or more engines. Coupled safety, above that of normal governing systems, is obtained from a simple, time-tested propeller-controlled hydraulic governor acting as a standby to the electronic system.

In addition to the thoroughly-proven Aeroprop, the control components can be installed within the engine nacelle and are interconvertible electrically. The result is an efficient and flexible



• Detailed information is provided in the Aeroprop Turbo Propeller Control Brochure, Sky AFPS 100. Write for a copy to your nearest distributor. If you need detailed installation information, ask us to send an engineer. For Aeroprop—ask General Motors Research Corp. with your Turbo Prop planning unit.



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AEROPRODUCTS DIVISION • GENERAL MOTORS CORPORATION • DAYTON, OHIO

AVIATION WEEK, May 26, 1960

"Give us the tools . . ."

The 81st Congress Can Halt the Administration's SOCIALIST PROGRAM

In his speech at Massachusetts Institute of Technology, Winston Churchill said that America's possession of the atomic bomb is all that has kept Soviet Russia from overrunning Europe and bombing London.

Our State Department knows that there has been another deterrent to aggressive warfare by Russia and a deciding one. That deterrent is the superior industrial strength of the United States. But once Russia approaches our industrial strength, then watch out! For Stalin or no Stalin, three will be trouble. Therefore, the simple table below is worth every American's careful reading. It shows in percentages what Russia did with her national income in 1948 and what we did with ours:

	RUSSIA	USA
Civilian use	60%	79%
New capital equipment and public works	21%	12%
Foreign aid		2%
Defense	13%	5%
Building of apartments and war stock-piling	6%	2%

These figures for Russia come from The (London) Economist, Britain's influential economic journal.

These figures are estimates based on information from behind the Iron Curtain, and so cannot be checked directly. But they fit with what is known of Russian development.

The table shows that Russia is striking every resource to build up its industrial strength. When Russia's effort is measured in dollars, and compared to ours, the figures show:

When we spent \$20 to \$21 billion for new industrial plants and equipment last year, the Russians spent \$12 to \$14 billion.

But while we used about \$9 billion of this to replace old equipment, the Russians spent no more than \$2 billion for replacing old equipment. The Russians had much less worn-out and obsolete equipment to replace. They could concentrate their efforts on expanding their industries and buying new equipment.

So—we used only \$11 to \$12 billion to expand our industries.

And the Russians used almost as much to expand theirs—\$10 to \$12 billion.

Russia is gaining industrial strength as fast as we are—and may soon be gaining faster. The more she gains and the faster she gains on us, the greater is the danger of war.

American industry is pushing modernization and expansion hard. It is doing an heroic job. McGraw-Hill's recent survey* shows that industry already has in hand plans to build plants and buy equipment in the next five years adding up to \$55 billion. Industry plans that investment—and much more—if it can get the money.

On these plans of industry depend our national security.

If these plans of ours are cut back, the Russians will be years closer to their goal of industrial equality—the strength that they need to wage aggressive war successfully.

But more and more our industry's plans are being nipped by socialist policies in Washington. The President continues to urge a further increase in the tax on corporate profits, even though federal taxes alone now take 38 cents of every dollar of profit. He wants \$3 billion more in taxes on corporate profits now, plus added personal taxes.

Last year corporations spent almost two-thirds of their profits—about \$13 billion—for new plant and equipment. This year corporation profits will be lower than last year's \$21 billion, perhaps by 20 per cent. Subtract a fifth or more from last year's profits. Then adopt the President's proposal and take \$3 billion more in corporate taxes and you have with planned expenditures for new plant and equipment.

Approval by Congress of the President's tax program would cut industry's program of plant and equipment development by a third or more. That means a major blow to our prosperity as well as our national security. For as capital investment goes, so goes general prosperity.

Further serious damage would be done by Congressional approval of the President's industry-control bill. The so-called Stability Act of

1949 (the Spence Bill) would severely check industrial progress. That bill would put the federal government in the business of providing the added industrial capacity which the tax program would prevent private industry from doing for itself. It would be hard to conceive a better and surer way to dry up private investment in new plant and equipment. For every dollar of government investment will scare away many times more dollars of private investment. People will not want to risk their money in businesses competing with the U. S. Treasury. At the same time it will attack private investment in another way. It means that government would spend your income for you instead of allowing you to spend or invest for yourself. That is the high and quick road to socialism.

American industry needs right now great courage and incentives if it is to carry out its tremendous building program. It needs also a release from the program of a socialist administration in Washington with its systematic discouragement of enterprise and risk taking.

Above all, industry needs assurance by the action of the 81st Congress itself that there is a future in this country for a system of dynamic capitalism, functioning in a free society. By acting now to strengthen the American people's faith in their industrial system, by providing needed incentives for management and invention, by protecting industry's capacity to buy new equipment, the 81st Congress can sustain American industrial progress and keep us settled and strong.

But if we kill freedom of industrial planning and action by increased taxes and government controls we put ourselves—and our friends all over the world—in dire peril.

Nothing would please the Communists more.

John H. McGraw, Jr.

President, McGraw-Hill Publishing Company, Inc.

*A complete report on our national survey may be obtained by writing McGraw-Hill Publishing Co., 330 West 42nd St., New York 36, N. Y. This is one of a special series of editorially selected results for new plants and equipment.

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SALES & SERVICE



UNIQUE FEATURES of Bellinger-Koppen plane: right slider, fold-up wings, bellows-edge ribs, rudder, control gear

New Slow-Flying Plane Developed

Craft is designed to fly at 30 mph. without stall, cruise above 100 mph., take off and land in 100 ft.

A lightplane with such low speed performance that it conceivably could open a new era in the personal aircraft field has been designed, built and test flown near Boston.

The plane is the result of secret development during the past year by Dr. Otto C. Koppen, noted aeronautical engineer of the Massachusetts Institute of Technology, and Paul Louis L. Bellinger of the Harvard Business School, notably known for research in aviation.

- **Slow Flight**—Although the test program is not yet completed, Aviators Wings has loaned the plane already has flown at approximately 27 1/2 mph. for six to seven several conditions.
- **Taken off** (lowest passenger) before the pilot could fully open the throttle.
- **Taken off** in about 60 ft. in still air with approximately 7 1/2 percent power.

This performance is better than some of the goals set by Koppen and Bellinger. Design objectives included producing a plane which could be slowed down to 50 mph. without risk of a stall, which could take off with full gear and no wind in 100 ft. or less and clear a 50-ft. obstacle less than 100 ft. from a standing start, which could land in 100 ft., and which would have a cruising speed substantially above 100 mph.

► **Helicopters**—Koppen and Bellinger do not maintain the plane has met at or

exceeded one of the design goals. They say they will make no performance claims until the test program has been completed with scientifically recorded data.

But they do state the plane will not stall and cannot be spun. Because it is intended for operation from the most crowded airports, and for a rotorcraft, they have named their craft the "Helio-plane," and have established the Helio Corp. The craft is expected to sell where its designers deem it ready for about \$300 more than conventional lightplanes.

► **New Wrinkles**—Personally because a number of innovations responsible for the plane's performance are patentable, the designers are withholding some details of construction. But the craft is full of new wrinkles and adaptations in new forms of previously known principles.

The mechanism which automatically counteracts a novel right slider is one of these features. Lower part of the rudder is linked to the rick control, making the plane essentially a two-control aircraft. Upper portion of the rudder is operated by conventional pedals.

A special oval-bladed constant speed, one foot propeller, with high thrust at low speeds, is another feature. It was built for the plane by Aeromatic Properties Department of Koppen Co.

Full span flaps and ailerons, already



NINE-FOOT propeller and Bellinger

naturally opened slots contribute to the slow flight, high lift characteristics. ► **No Stall, No Spin**—"It is a genuinely stall proof airplane," Bellinger told Av-

ANON: Where "Since it is completely still and dead, it is also very good." At 30 mph the wing is 14 deg. thus stalling angle—farther from a stall than a C-130 in normal flight."

It is a quiet airplane with an elegant profile and the prop geared down. The system here, attempted to make it as innocuous as the airplanes developed under NASA sponsorship by the Aeronautical Research Foundation and demonstrated in Boston last summer. Although both men helped develop the foundation's research aircraft and sought properly have used the same type of four-blade propeller, Kappan do speak an entirely new arrangement.

► **Vagabond Start-Up**—When they started the work more than a year ago Kappan and Bellinger had intended merely to modify a Piper Vagabond. Before they came through, nearly everything on this plane was new. Except for the cabin floor, tubing, gear, and windshield and rubber seals. "It's all new from the cabin back, and from the firewall forward," Bellinger explained.

In the work program, the single door was discarded, there is now one on each side. About 45 cubic feet have been added to the volume length. The span is about 28 feet (7 inches less than a Piper Cub). It has Gossamer two-stroke gear, starter, generator, battery and fuel tank.

Powerplant is an 85-hp Continental with fuel injection. A tank of about 20 gal. capacity was installed behind and below the side-by-side seats.

Other features: wing area, approximately 150 sq ft; maximum allowable gross, 1150 lb (91 lb overload for heavier loads); empty weight, about 550 lb; length, about 50 ft; useful load, wing loading, about 10 lb per sq ft.

Incorporated in the loading gear, which has been moved forward to a new, permanent-alter arrangement. The fuel tank, the equipment, however, are so distributed that no conventional light-plane. Since this part of the airplane was considered "optional," they give it no special engineering attention—and it is the only part of this model aircraft that has given mechanical trouble to date. A new instrumentation arrangement is being installed.

► **More Coming**—Kappan and Bellinger believe they have created a new type of aircraft, one that is a significant advance in aircraft design, especially in light aircraft, slow speed aircraft suitable for personal flying, agricultural work and military liaison. Development work is believed to be less than one-half that of a small aircraft factory would have spent.

They expect to have a small number of these airplanes assembled in the Boston shops of E. W. Wiggin Airways, to be used for service and purposes. Only after the design is fully developed and thoroughly tried will they be looking to



LEADING EDGE SLATS, automatically retract, contribute to plane's high lift.

increase it for transportation and sale to the general public. Next steps: completion of test data and obtaining an Approved Type Certificate from CAA.

Renewing the project, Bellinger told AVIATION WEEK.

► **The Why**—We were convinced that the small airplane was failing almost completely, to meet the real purpose for which it was put on the market, by being or more willing to buy. The total market had been small. Potential customers don't buy for sport any more, they buy for transportation.

With nothing except, it has been demonstrated that 180 cubic feet is generally about the maximum volume which can be flown to one hour. But on a trip of that distance it is cheaper to use the airplane. Secondly, as a trip of that length, the volume becomes too impractical. Then there has been the lack of safety in the general airplane.

"We were convinced that to solve the problem there would have to be a complete overhaul of the technology, making the aircraft as safe as a modern day car."

The first objection to such close as aspects was none. The Aeronautical Research Foundation's project on more under NASA's sponsorship demonstrated that it could be discarded. We wanted a rational model, again, in Boston, to be located in quiet places. Although the Massachusetts General Court heavily reduced the project and passed enabling legislation in 1947 granting permission to use certain public property, the project later was blocked by conflicting interests.

The quiet airplane was one step—no important step. Otto Kappan had been working for several years on ideas for shortening the space required for landing and takeoff. It seemed to me that his solution was more ingenious, effective and simpler than anything else. I tried—we both tried—to get the

Civil Aeronautics Administration to sponsor the project. T. P. Wright, then administrator, was unable to obtain funds from Congress. So Kappan asked his design to the Aeronautical Research Foundation.

The Foundation agreed as to possible sponsorship by the National Aeronautics Administration. Kappan's representative advised that such a complete airplane project was not paper for them. The Foundation itself had all the work it could handle and it seemed, as its trustees were also forced to turn down this project.

► **Pin the Vagabond**—Unable to obtain a design, or public support, Kappan and Bellinger at that point decided to try to develop the airplane of their own risk as a private venture. They had mechanical and Wiggins Airways in New Bedford, Boston. Kappan had the last engineering foundation and his associates helped him. Bellinger set out all specifications, based on his appraisal of market needs.

The objective was to maintain the existing cruising speed, payload and power requirements of the present day aircraft, and in addition to be able to fly safely in low maintenance conditions and to be able to land and take off on less than 100 feet.

A major difficulty which was encountered as all attempts to a time limit, ultra-violet light with fuel usage has been the extreme difficulty of maintaining stability and control.

On the question of this plane's performance, Bellinger comments: "If Kappan's design can take off on less than 100 feet with only 85 hp and our maximum gross weight of less than 1000 lb, a revolutionary technical advance will have been made in light plane performance."

The make story, of course, is that the performance or better has already been achieved with the plane.

THE CITIES SERVICE FLIGHT TEST

Power Prover

This remarkable instrument indicates the combustion characteristics of any type of engine and atmospheric conditions in related engine parts.



The Flight Test Power Prover



is typical of Cities Service's contribution to better flying. This instrument indicates combustion and atmospheric conditions in all types of engines and related equipment where atmospheric investigations are required. Through its use many engines' operating problems can be solved.

The Cities Service Flight Test Power Prover helps improve helicopter operation, resulting in smoother engine performance and better handling of fuel.

Cities Service has grown up with aviation. It has pioneered in many unique and unusual products and services for better flying. This ex-

perience stands solidly behind the complete line of top quality aviation products of Cities Service. Look for the green and white aviation emblem at more and more airports every day.

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New Idea in Lightplanes

Contributions are an index for Dr. Otto C. Koppert of Massachusetts Institute of Technology, Prof. Lynn Bellinger of Harvard Business School, and their group of associates and stockholders in the Hula Corp.

Elsewhere in this issue, we report the somewhat perfect way already achieved in preliminary tests of the Heliplane, a revolutionary two-place, sport, non-ops, non-sail craft. Takeoff and landing, says it will under 100 feet. Although cruising speed of more than 100 mph will be possible, the plane can fly 16 mph safely.

The first Heliplane is scheduled at a crucial stage of personal aviation. Lightplane sales are off. Procs are gone. Flight schools and fixed base agencies are going out of business in substantial numbers. The industry needs new and strong ideas desperately, but standard business school strategies such as an airplane Heliplane provide serious emergency like the Hula Corp. offer one of the answers to the demand for personal airplanes that reflect the latest technological developments.

Such a project also may become a gamma jet in modernizing the long abandoned requirements of the Civil Aeronautics Administration for approval type certificates which in the past have acted as a barrier to private aircraft type in the specifications in the original certificates. No company has ever been able to add all of the new features it wished because it could not afford the complete testing CAA demanded for another certificate.

Even though the Heliplane being discussed this week still lacks some refinements its sponsors hope to introduce in their efforts to create a high-altitude, side plane, as well as a new design. The Heliplane—even in its current test stage—appears to be the most important new idea in lightplanes introduced since the previous Europe.

Ignorance Is Not Bliss

Several active members who have appeared before Sen. Johnson's subcommittee have left the stand amazed at the lack of knowledge members of Congress have during an aviation.

The United States Air Force met a similar problem and failed it. It was managed an effective aviation program on Capitol Hill and has made unusually good use of its dramatic accomplishments. The Air Transport Act, on the other hand, has been off so long on any effective, detailed information program that it is beginning to imp the inevitable result.

United Air Lines' President W. A. Patterson testified Sen. Johnson by declaring that in the more than ten years the Civil Aeronautics Act has been in effect, not one member of the Civil Aeronautics Board, or one of its staff, had ever visited the active headquarters to study United's methods.

This is a sorry state of affairs. It answers a clue to some of the unexplained, prolonged decisions of the Board in ways past. But it also makes one wonder what has happened to management's sense of public relations. It is industry's obligation to publicize its own facts of life, honestly and

fairly and completely. Ignorance breeds on misinformation or lack of information.

But airline management has failed to realize the importance of industry-coordinated public relations. It has, instead—d it gave any thought to the matter at all—devoted its public relations resources to stress only company developments and to bludge with the newspaper. The result has been confusion for Congress, the public, and the Government, and a serious net loss in air transportation generally. Facts are easily taught by everybody, but not leadership, suggestions, bold tries, and occasional saving of individual company statistics, ability and special pleading.

It is time for more cooperative goal setting and status-keeping in air transportation public relations.

Fly the Masses

The airline parasites line up one by one saying that air is coach service.

Mr. 28 we quoted Pan American's John Tupper: "The people want tourist class air service, and it is late to stay."

On Apr. 14, we quoted Capital's pioneering J. H. Casanovi: "Air coach service is profitable."

Now, Northwest's Civil Hatter tells the Johnson room mates of the Senate: "It is my opinion the present passenger fares of the airlines have put them out of the mass travel market. Before the war our average passenger fare was slightly under 45 cents per passenger mile. I believe passenger fares at this level would increase as travel demand increases to such an extent that profitable bus factors would be situated on all routes represented—on but, on all equipment currently being flown by the airlines, except the DC-3."

Hatter says further: "There is no reason why the airlines should not provide several classes of service with different fare levels. Present records indicate that there is a travel market sufficiently large to support a reasonable number of heavy flights on non-stop and limited schedules with most air generated but equipment." The lower coach rate will attract new business that otherwise may not travel at all or that which will be taken from automobiles, buses, and cheaper train service. In the last two years have been created in a year when the public was becoming very cost-conscious and the chart of the volume of passenger traffic shows clearly that when air fares were put into effect, the volume of air travel started to go down."

Meanwhile, as excitement grows within the industry, CAB turns down in delay new route fare applications, fearful of a heavier competitive situation.

Some frequency reduction is justified, and for a time it may be necessary to prohibit coach service with the most modern aircraft. But as so far has been the only means of mass aviation that has been the public has traveled in luxury and comfort at all. It is only a matter of time until the masses will demand a good service at second class fares, and the airlines and CAB must bow to it. Someday, just as in the airlines today, luxury schedules will be in the minority. But before that happens some segments of the industry will have to give the fat out of their income costs to give up the ghost completely to airline competition.

ROBERT H. WOOD



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